

Effects of asteroids on the orbital motions of terrestrial planets

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Abstract

The present planetary ephemerides, as INPOP08 (Fienga et al., 2009) [1] and DE405 (Standish et al., 1998) are subject to a lack of accuracy, because of the perturbations arising from a large number of asteroids. Those perturbations could reach a few kilometers in several decades in the case of Mars for instance. So it seems appropriate to study in details the individual specific effects of these asteroids.

As an illustration this work deals with the evaluation of the individual effects of largest asteroids of the solar system on the orbits of the terrestrial planets Mercury, Venus, Earth and Mars.

Our methodology consists of several steps:

- A numerical integration of the orbits of the planets with and without the disturbing asteroid from which we want to know the effects.
- A determination of the signal representing the effects, by simple subtraction.
- The analysis of the signal by the method of
 - FFT (Fast Fourier Transform) to determine the most significant sinusoidal oscillations.
 - Adjustment of the signal by the set of sinusoids determined in the previous step.

This type of study is interesting in many fields, such as planetary ephemerides, as well as spatial navigation, to understand better the effects of each asteroid taken individually on the terrestrial planets. Note that this type of study is a continuation of previous studies (Williams, 1984 [3]; Mouret et al., 2009 [2]).

Introduction

The motion of a given planet around the sun can be considered at first approximation as a Keplerian motion perturbed by the other planets and the small bodies of the solar system. Each of these perturbations must be treated either analytically or numerically, and can be measured as a change of the planet's osculating orbital elements $(a, e, i, \Omega, \varpi = \Omega + w$ and $L = \varpi + M)$ determined from the perturbing function \mathfrak{R} , according to Lagrange's formula

$$\begin{aligned} \frac{da}{dt} &= \frac{2}{na} \frac{\partial \mathfrak{R}}{\partial L} \\ \frac{de}{dt} &= -\frac{\sqrt{1-e^2}}{na^2 e} \left(1 - \sqrt{1-e^2} \right) \frac{\partial \mathfrak{R}}{\partial L} - \frac{\sqrt{1-e^2}}{na^2 e} \frac{\partial \mathfrak{R}}{\partial \varpi} \\ \frac{di}{dt} &= -\frac{1}{na^2 \sqrt{1-e^2} \sin i} \left[\frac{\partial \mathfrak{R}}{\partial \Omega} + (1 - \cos i) \left(\frac{\partial \mathfrak{R}}{\partial \varpi} + \frac{\partial \mathfrak{R}}{\partial L} \right) \right] \\ \frac{d\Omega}{dt} &= \frac{1}{na^2 \sqrt{1-e^2} \sin i} \frac{\partial \mathfrak{R}}{\partial i} \\ \frac{d\varpi}{dt} &= \frac{\sqrt{1-e^2}}{na^2 e} \frac{\partial \mathfrak{R}}{\partial e} + \frac{1 - \cos i}{na^2 \sqrt{1-e^2} \sin i} \frac{\partial \mathfrak{R}}{\partial i} \\ \frac{dL}{dt} &= n - \frac{2}{na} \frac{\partial \mathfrak{R}}{\partial a} + \sqrt{1-e^2} \left(\frac{1 - \sqrt{1-e^2}}{na^2 e} \right) \frac{\partial \mathfrak{R}}{\partial e} + \frac{1 - \cos i}{na^2 \sqrt{1-e^2} \sin i} \frac{\partial \mathfrak{R}}{\partial i} \end{aligned}$$

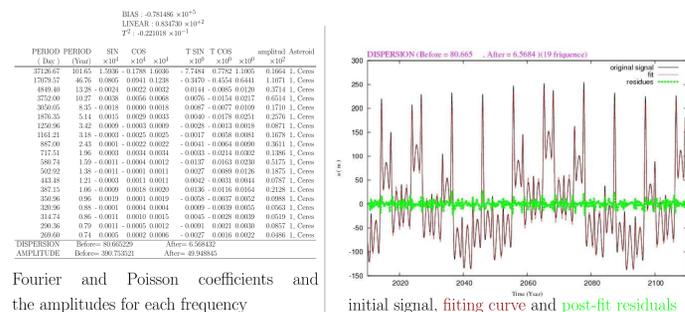
To evaluate the effects of a given asteroid on the terrestrial planets we use the numerical integration (Runge-Kutta of the 12th order), in the frame of the 9-body problem (the Sun and the eight planets without asteroids), then of the 10-body problem (the Sun and the eight planets together with the given asteroid). Then we determine the differential variations of orbital parameters of the planet by simple subtraction of the two signals obtained. After we perform the frequency analysis of the data, using fast Fourier transform (FFT) to determine the leading frequencies. At last we carry out a nonlinear regression in which the differential data are modeled by least-square method following an equation of type :

$$F(t) = \sum_{i=1}^N A_i \sin(f_i t) + B_i \cos(f_i t) + C_i t \sin(f_i t) + D_i t \cos(f_i t)$$

We have also calculated the individual influences of each asteroid on the distance from the EMB (Earth-Moon barycenter) to the given planet and the orientation vector of this planet as seen from the EMB, which are very important parameters in space navigation and astrometry.

We present below an example of our results which consist in tables showing the coefficients of Fourier and Poisson components for the orbital elements of each terrestrial planet with respect due to each asteroid, and the corresponding curves (the initial signal, the adjustment determined by our analysis and the residuals). In each case: (planet, asteroid, orbital element) we find that our fit is satisfactory, since the post-fit residuals are significantly lower than the original signal.

Gravitational influence of Ceres on the semimajor axis of Mars



Gravitational influence of Ceres on the EMB-Mars distance

